



*Idaho National Engineering and Environmental Laboratory*

# ***RELAP5-3D Compressor Model***

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# Compressor characteristics

- *Positive Displacement*
  - *Low flow rates, high pressure ratio*
- *Dynamic*
  - *Convert velocity to pressure in continuous flow process*
  - *Centrifugal*
    - $1.3 < P_o/P_i < 13$
    - $75\% < \eta < 87\%$
  - *Axial flow*
    - $1.1 < P_o/P_i < 1.4$
    - $80\% < \eta < 91\%$

# Centrifugal Compressor

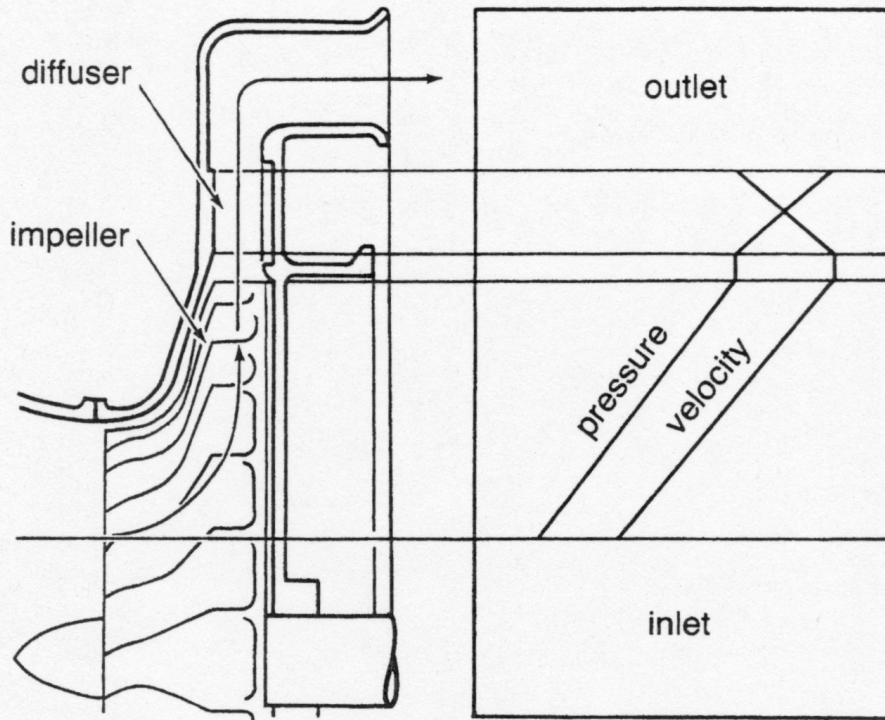
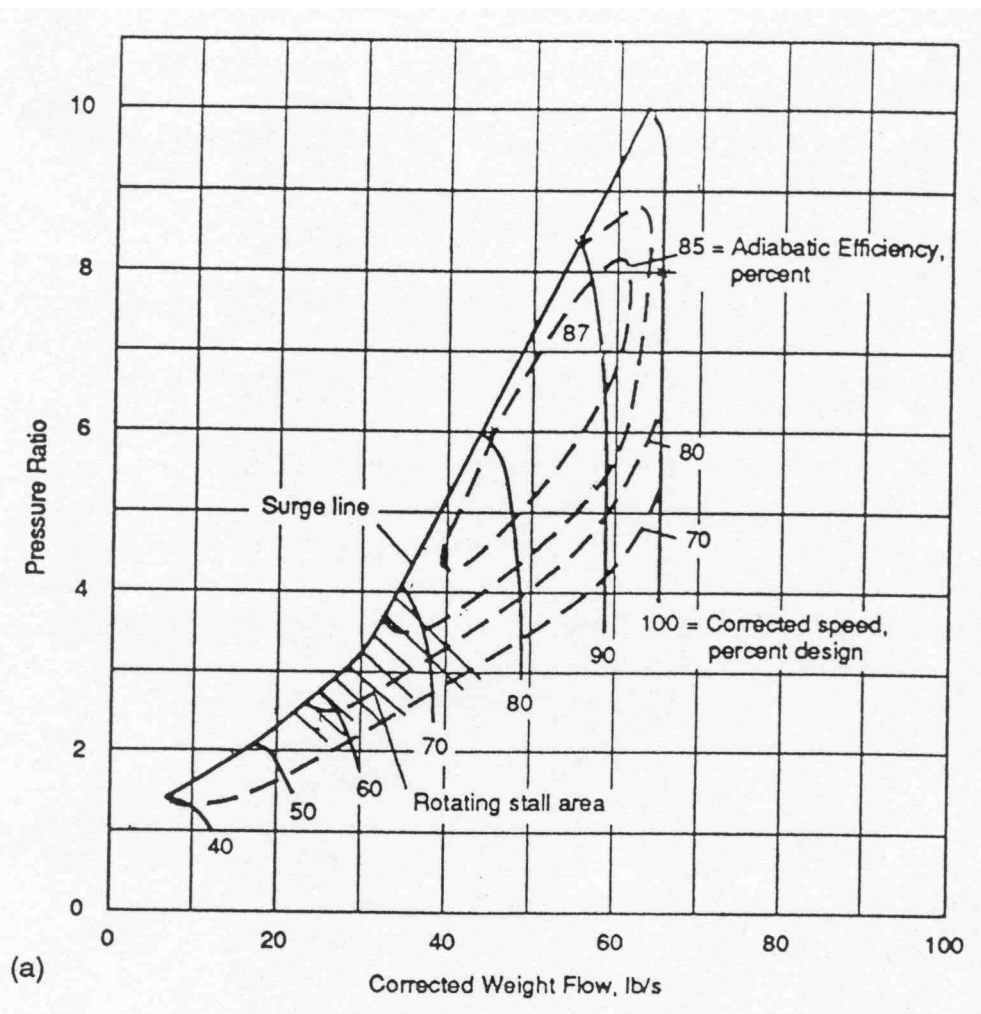


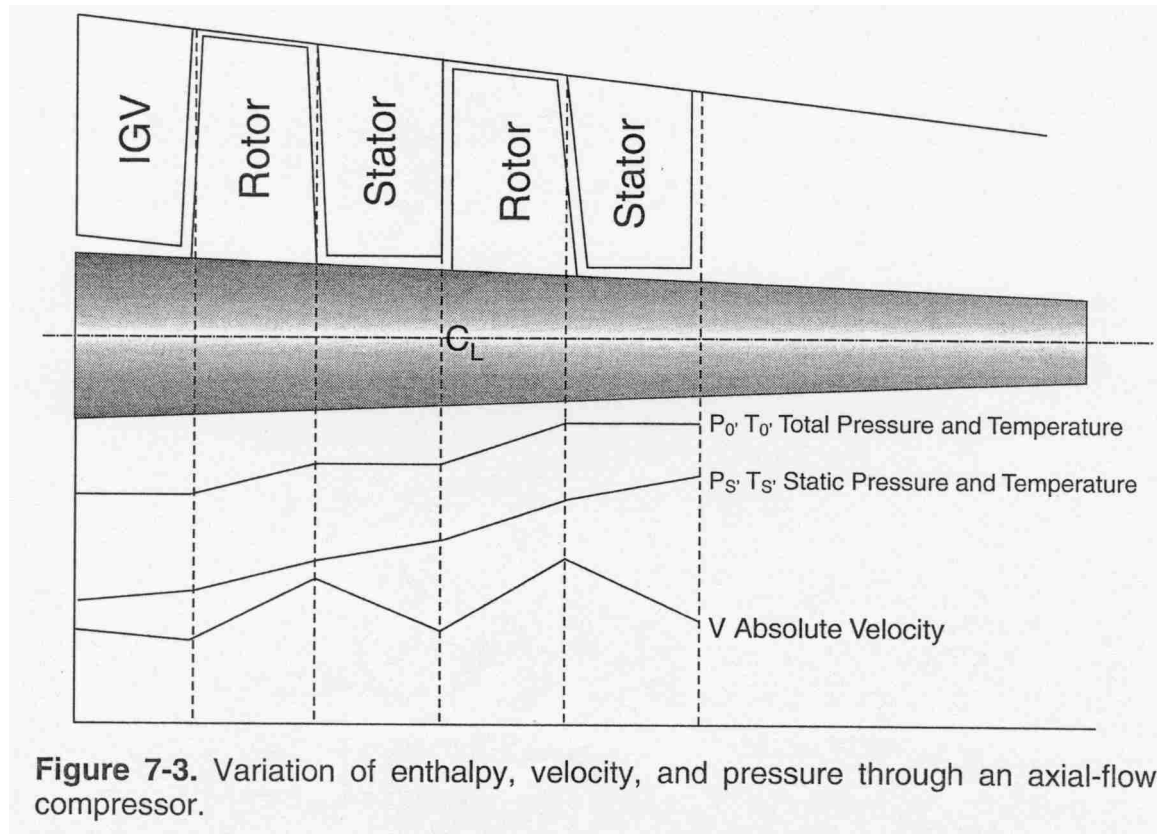
Figure 6-1. Pressure and velocity through a centrifugal compressor.

# Centrifugal Compressor Performance Characteristics

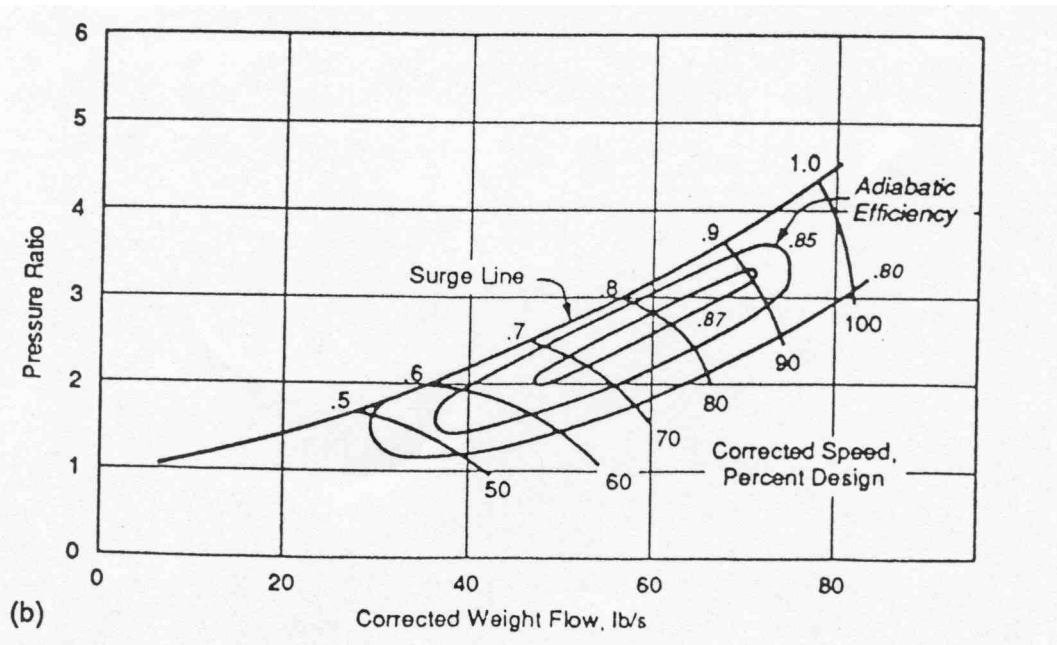


(a)

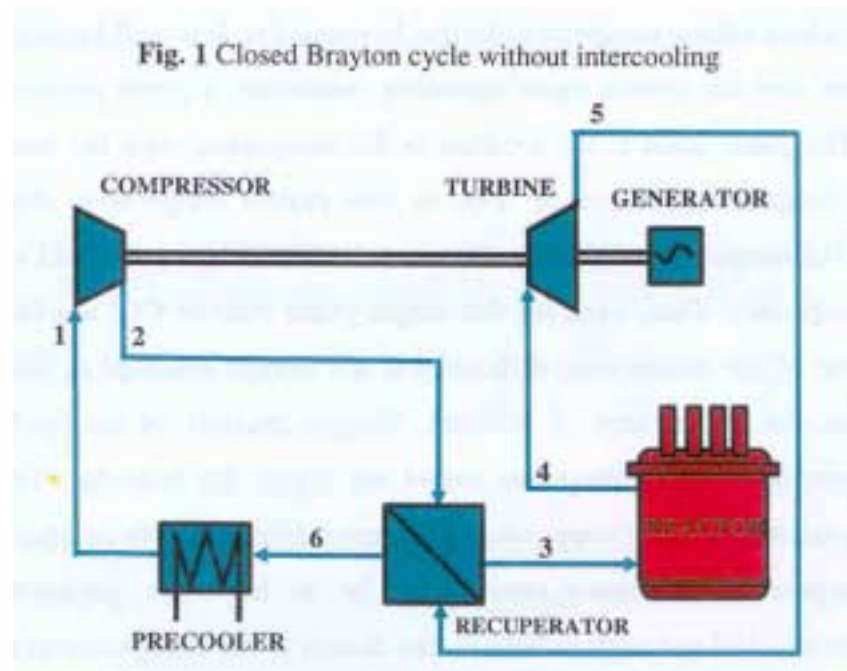
# Axial Flow Compressor



# Axial Flow Compressor Performance Characteristics

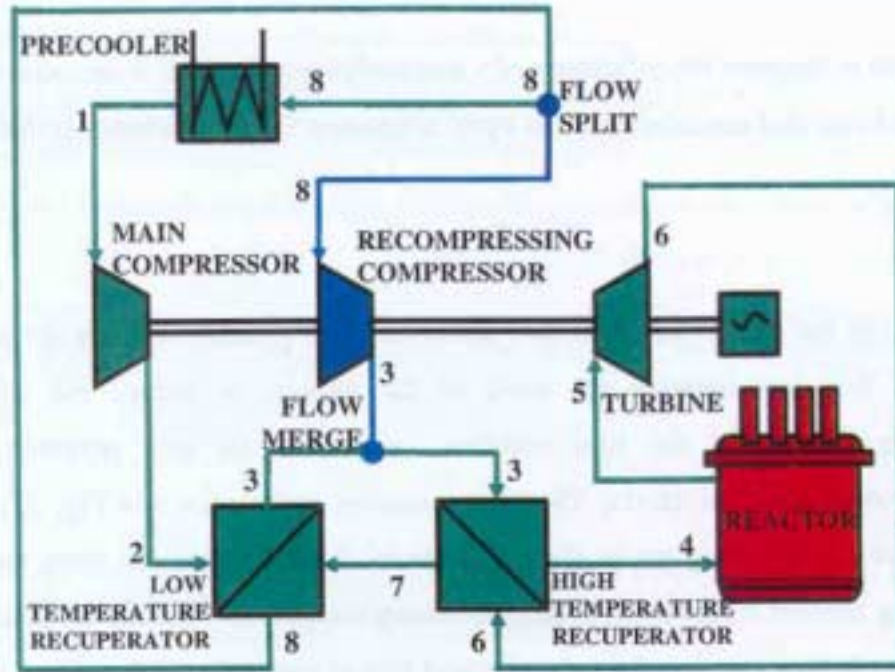


# Simple Brayton Cycle Reactor



# Recompression Brayton Cycle Reactor

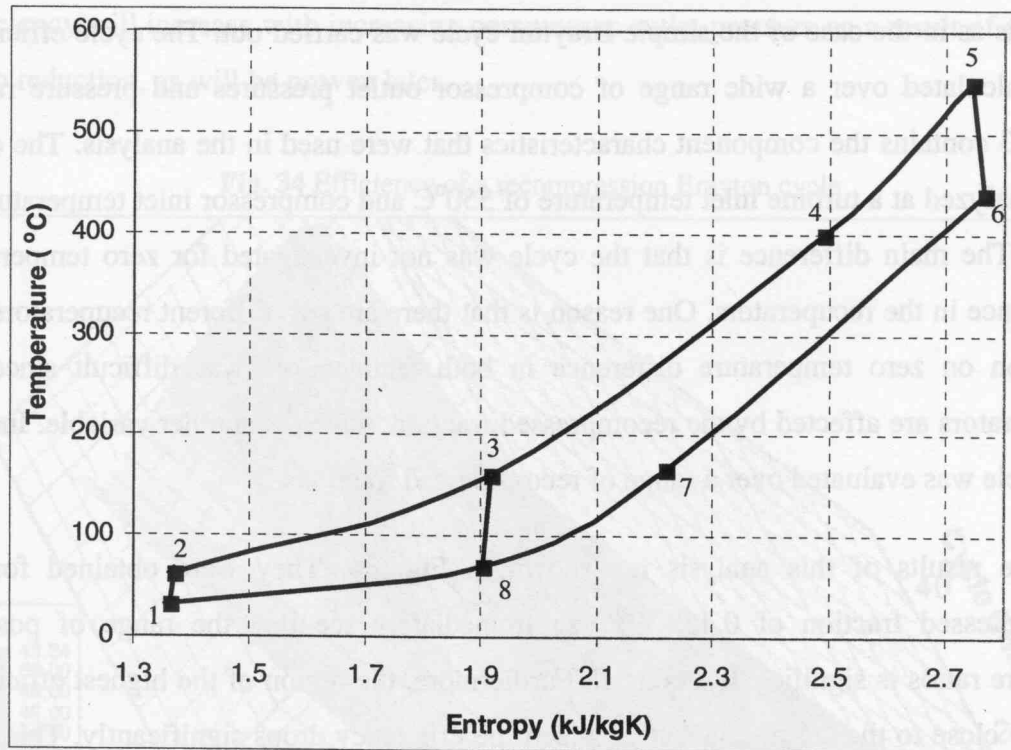
Fig. 32 Recompression Brayton cycle layout





# Recompression Brayton Cycle Diagram

Fig. 33 Temperature-entropy diagram of a recompression Brayton cycle



# ***A compressor is similar to a pump***

- *Rotational Velocity*
  - *Input from table with or without trip*
  - *Torque-Inertia equation, optionally with motor torque*
  - *Shaft rotational velocity equation*
- *Spindown (coastdown) Data*
- *Dissipation*
- *Rotor Inertia*
- *Configuration differences*
  - *Inlet junction has head added to fluid*
  - *Volume is outlet state*
  - *Optional outlet junction*
  - *Outlet can be connected to another compressor or a non-compressor*

# Performance Characteristics

- *Normal Operation*
  - *Region between the surge and choke points*
  - *Surge*
    - *aerodynamic instability in impeller or diffuser*
    - *intermittent flow direction (and force direction) reversal*
  - *Choking*
    - *sonic flow at minimum area point*
    - *efficiency drops rapidly*
- *Reverse flow*
  - *Turbine*
  - *Further investigation pending development of design information*
- *Reverse direction*
  - *Further investigation pending development of design information*

# Compressor H-S Diagram

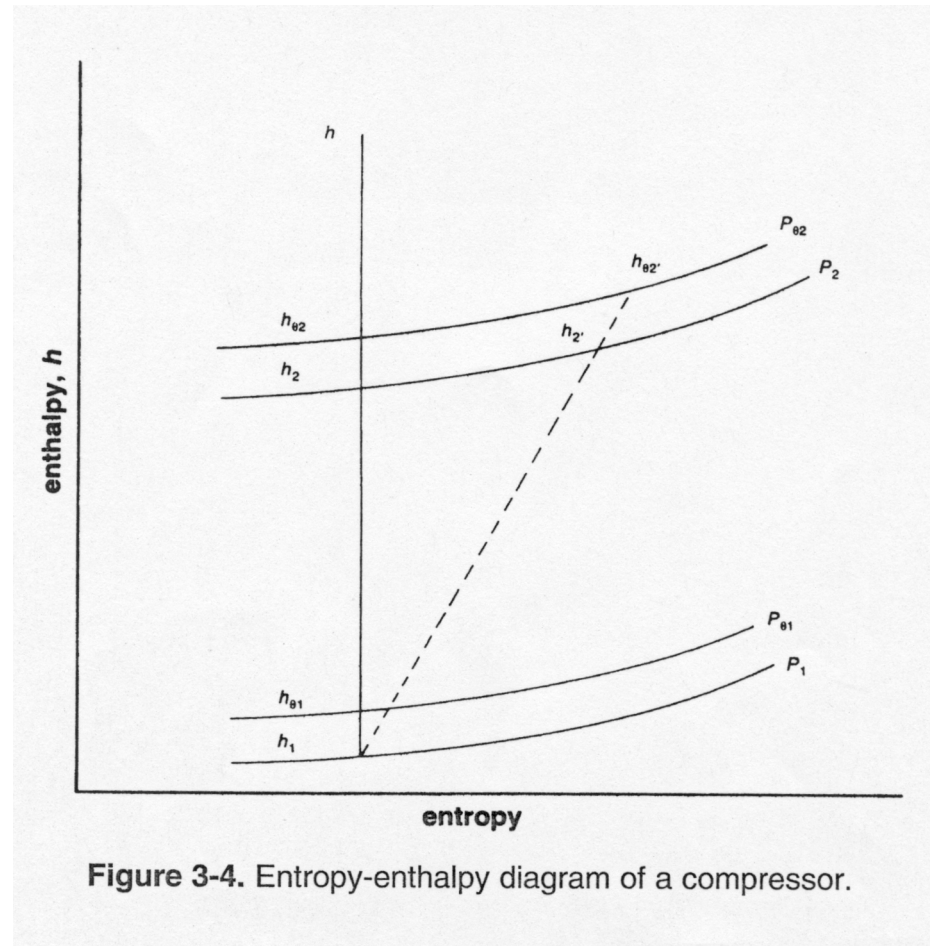


Figure 3-4. Entropy-enthalpy diagram of a compressor.

**A change in angular momentum of working fluid is caused by tangential forces (only). Isentropic torque can be calculated by considering an isentropic compression of the fluid ...**

$$\tau = \dot{m} (r_1 \cdot V_{\theta 1} - r_2 \cdot V_{\theta 2}) \quad (1)$$

$$\tau \cdot \omega = \dot{m} (r_1 \cdot \omega \cdot V_{\theta 1} - r_2 \cdot \omega \cdot V_{\theta 2}) \quad (2)$$

$$\dot{Q}_{c.v.} + \dot{m} \left( h_1 + \frac{V_1^2}{2 \cdot g_c} + Z_1 \frac{g}{g_c} \right) = \dot{m} \left( h_2 + \frac{V_2^2}{2 \cdot g_c} + Z_2 \frac{g}{g_c} \right) + \dot{W}_{c.v.} \quad (3)$$

$$\dot{W}_{c.v.} = \tau_s \cdot \omega = \dot{m} (h_1^T - h_2^T) \quad (4)$$

$$s_1 = s (h_1^T, \rho_1) \quad (5)$$

**The work in the azimuthal direction (only) is isentropic, so an isentropic outlet state can be defined.**

$$R_P = R_P(\omega, \dot{m}) = \frac{P_2^T}{P_1^T} \quad (6)$$

$$P_2^T = P_1^T \cdot R_P \quad (7)$$

$$s_2 = s_1 \quad (8)$$

$$h_2^T = h(P_2^T, s_2) \quad (9)$$

$$\rho_2 = \rho(P_2^T, s_2) \quad (10)$$

**Real outlet state is then obtained from definition of efficiency.**

$$\tau_s = \frac{\dot{m}}{\omega} (h_2^T - h_1^T) \quad (11)$$

$$\eta_{ad} = \frac{\text{Isentropic work}}{\text{Actual work}} = \frac{h_2^T - h_1^T}{h_{2'}^T - h_1^T}, \quad (12)$$

$$h_{2'}^T = \frac{h_2^T - (1 - \eta_{ad}) \cdot h_1^T}{\eta_{ad}} \quad (13)$$

$$\tau_d = \frac{\dot{m}}{\omega} \frac{1 - \eta}{\eta} (h_2^T - h_1^T). \quad (14)$$

$$\dot{W}_d = \tau_d \cdot \omega \quad (15)$$

**If assumption is made about outlet state, actual work done on the fluid can be calculated without entropy-based property table lookup call.**

$$\int_{P_1}^{P_2} \frac{dP}{\rho} + \frac{V_2^2 - V_1^2}{2} = g \cdot H, \quad \text{Assume: } \rho_m = \frac{\rho_1 + \rho_2}{2} \quad (16)$$

$$P_2 + \frac{\rho_m V_2^2}{2} = P_1 + \frac{\rho_m V_1^2}{2} + \rho_m \cdot g \cdot H \quad (17)$$

$$P_2^T = P_1^T + \rho_m \cdot g \cdot H = P_1^T + P_1^T (R_P - 1) \quad (18)$$

$$\Delta P = \rho_m \cdot g \cdot h = P_1^T (R_P - 1) = P_2^T \left( \frac{R_P - 1}{R_P} \right) \quad (19)$$



**Use efficiency to separate isentropic and dissipative components of torque.**

$$\dot{W}_{c.v.} = \dot{m} (h_{2'}^T - h_1^T) = \dot{m} (h_2^T - h_1^T) + \dot{m} (h_{2'}^T - h_2^T) = \dot{W}_s + \dot{W}_d \quad (20)$$

$$\dot{W}_s = \dot{m} \cdot g \cdot H = \frac{\dot{m} \cdot \Delta P}{\rho_m} \quad (21)$$

$$\tau_s = \frac{\dot{m}}{\omega} (h_2^T - h_1^T) = \frac{\dot{m}}{\omega} \frac{\Delta P}{\rho_m} \quad (22)$$

$$\tau_d = \frac{\dot{m}}{\omega} \frac{1-\eta}{\eta} (h_2^T - h_1^T) = \frac{\dot{m}}{\omega} \frac{1-\eta}{\eta} \frac{\Delta P}{\rho_m} \quad (23)$$

## ***Head and Torque Calculation Summary***

- *Isentropic torque can be derived using first principles*
  - *Pressure ratio and inlet entropy determine isentropic outlet state*
  - *Efficiency determines real outlet state and dissipative torque*
- *RELAP5-3D implementation*
  - *Pressure Ratio, linearized density used to calculate real outlet state*
  - *Hand calculations verify the accuracy of linearized density assumption*
  - *Efficiency determines isentropic and dissipative torque components*
- *Dissipative torque added to energy eqn*

# ***Performance Data Issues***

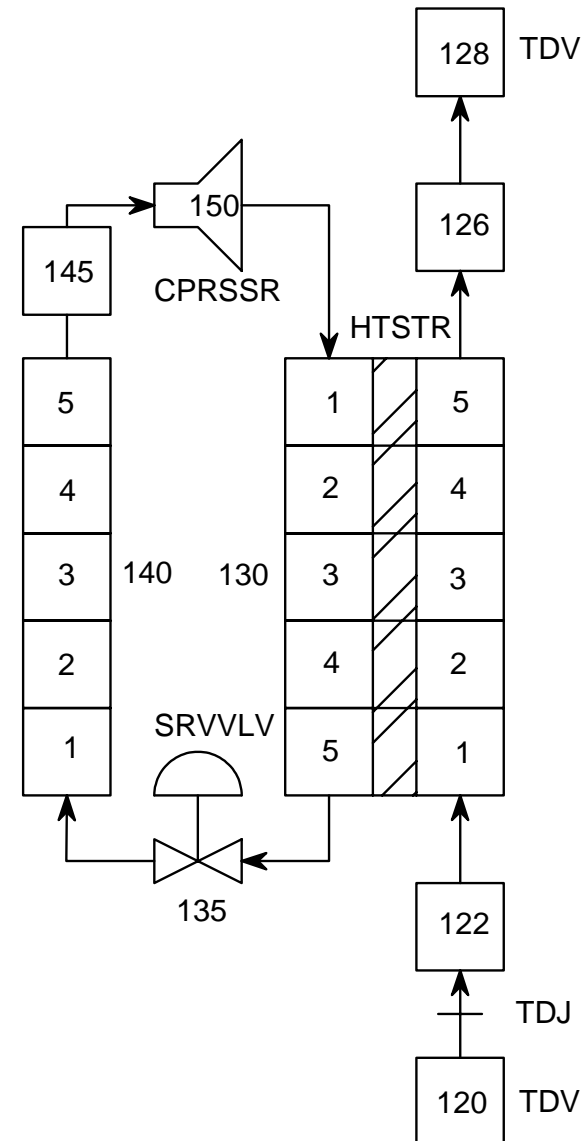
- *Surge line limit*
  - *startup*
  - *transient response*
  - *input from designers*
  - *gain experience with test problem*
- *Reverse flow, reverse direction, spin-down*
  - *input from designers*
  - *Incorporate capability for input*

# ***Implementation Status***

- *Input processing (rlevel) and cross checking (ilevel) completed*
- *Compressor model completed with momentum and dissipation terms in semi-explicit method*
- *Test model built*
  - *Simple loop with compressor, heat exchanger, orifice*
  - *Performance data from automobile turbocharger compressor*
  - *Extrapolation and mesh refinement to provide necessary numbers*
- *Component variables added to rstplt file*
  - *Cprvel, cprhead, cprtrq, cpreff, cprmt, cprnrt*

## Test Input File

- *Simple loop with heat removal*
- *Working fluid is air at atmospheric pressure*
- *SRVVLV component fully open*
- *Wall Friction in Pipe 140*
- *Calculations*
  - *Ramp from zero to rated speed*
  - *Steady state at rated speed*

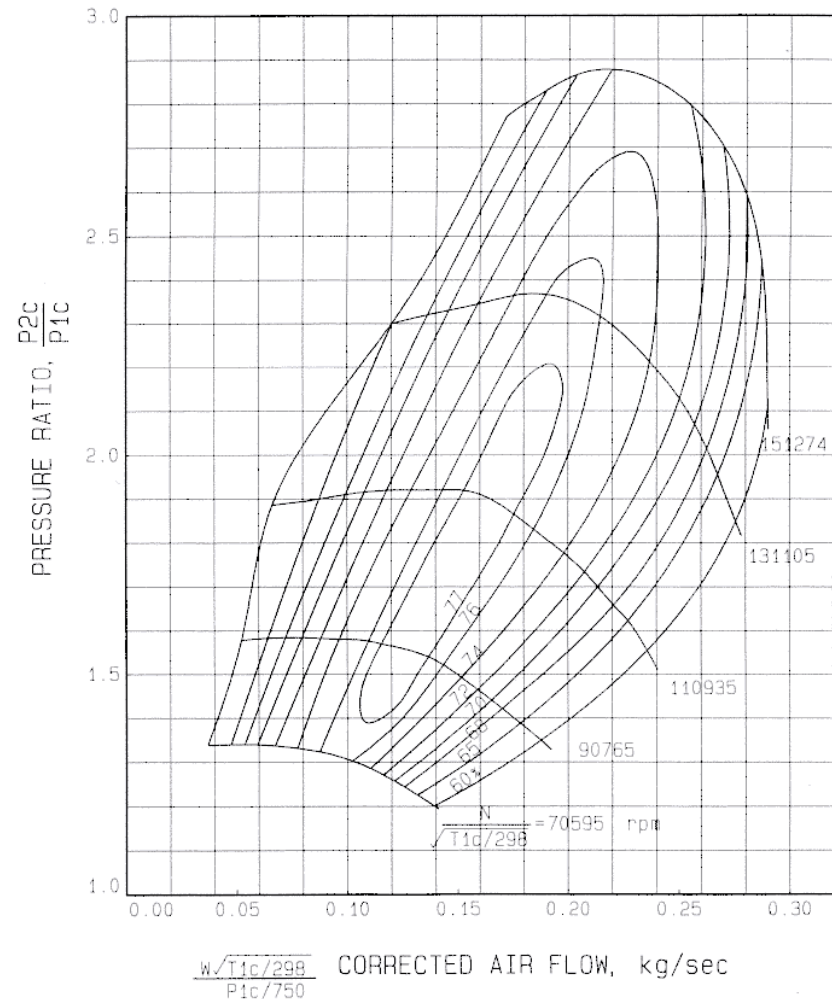


COMPRESSOR PERFORMANCE  
MODEL TD05-16G

"Small" compressor wheel  
49178-05200

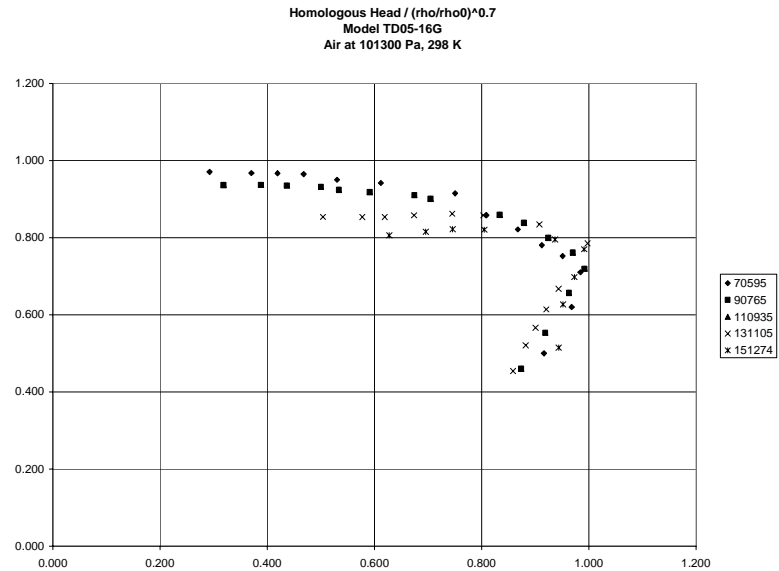
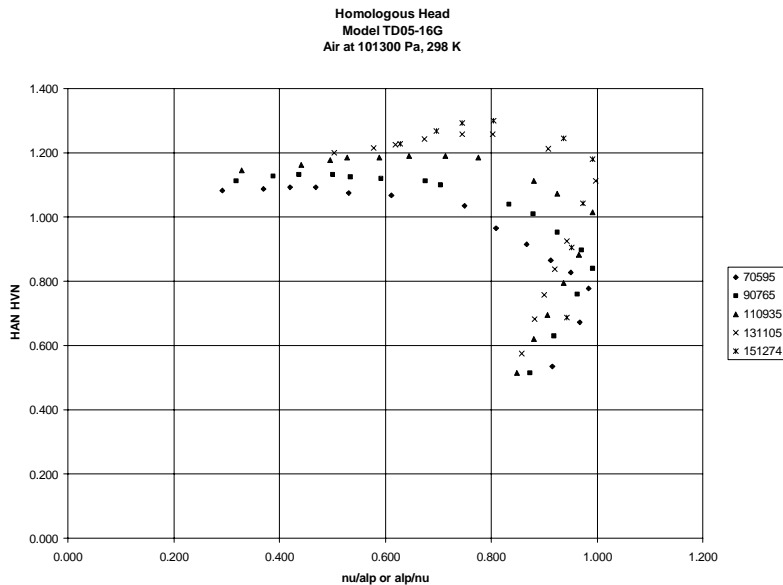
# Automobile Turbocharger Compressor

- Centrifugal Compressor
- Air as working fluid
- Data digitized
- Homologous form



mitsubishi heavy industries, ltc

# Homologous Representation of Data (air at atmospheric conditions)

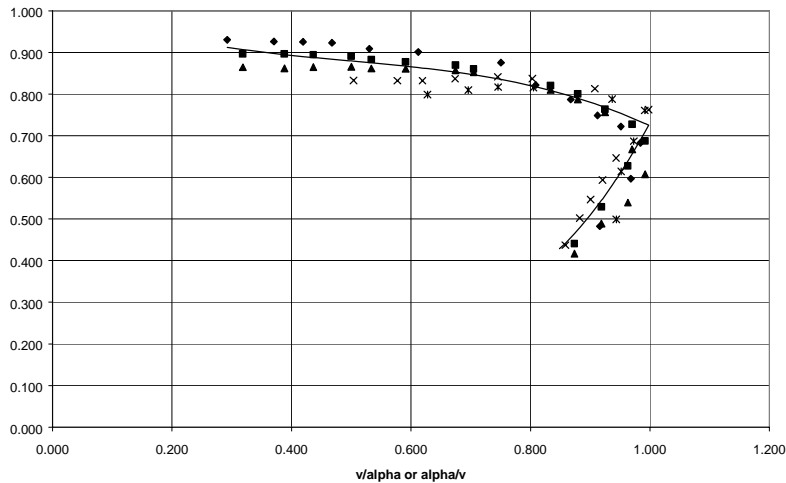


Uncorrected

$$H_c = H \cdot \left( \frac{\rho_0}{\rho} \right)^{0.7}$$

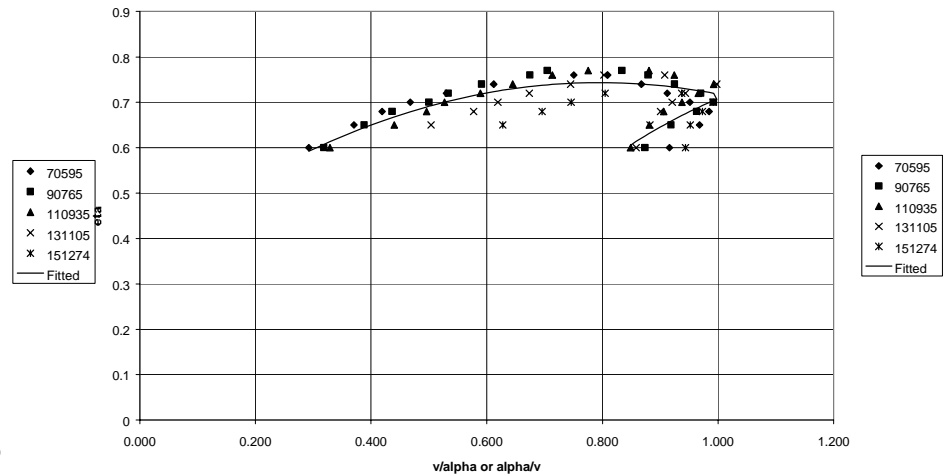
# Homologous Representation of Data (Supercritical CO<sub>2</sub> at 7.6 MPa, 305K)

Homologous Head /  $(\rho/\rho_0)^{2.2}$   
Model TD05-16G  
CO<sub>2</sub> at 7.6 MPa, 305 K



Head

Adiabatic Efficiency  
Model TD05-16G  
CO<sub>2</sub> at 7.6 MPa, 305 K



Efficiency

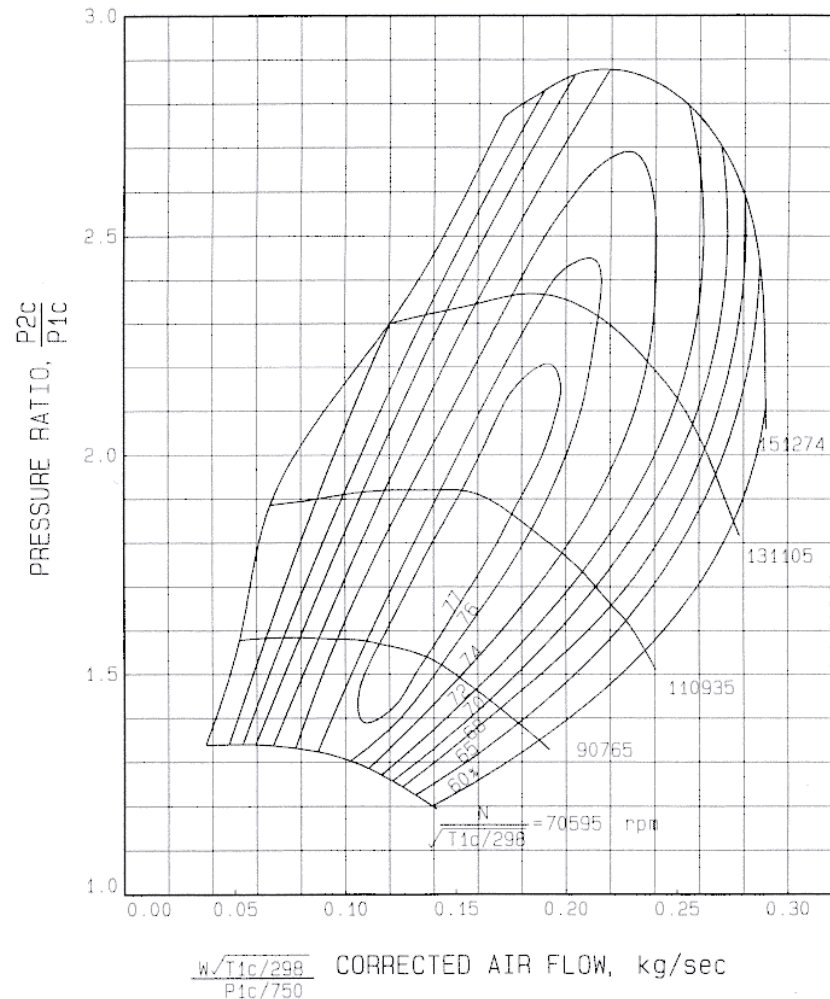


# Accuracy Comparison of Simple Linear Interpolation and Homologous Curves

- Estimate values for 11093 rpm speed curve.
- Simple linear interpolation between 90765-rpm and 131105-rpm curve at various flow rates
- Compare with density-corrected homologous representation
- Comparison made for Pressure Ratio and Efficiency

COMPRESSOR PERFORMANCE  
MODEL TD05-16G

"Small" compressor wheel  
49178-05200



MITSUBISHI HEAVY INDUSTRIES, LTD

## **Results show that simple interpolation is better than homologous representation for head**

$$\text{Fractional Error} = \frac{|\text{calculated} - \text{actual}|}{\text{actual}}$$

<i>Flow Rate (kg/s)</i>	<i>Pressure Ratio</i>	<i>Interpolated</i>	<i>Fractional Error</i>	<i>Homologous</i>	<i>Fractional Error</i>
0.12	1.920	1.933	0.007	1.923	0.002
0.14	1.923	1.918	0.003	1.907	0.008
0.16	1.910	1.903	0.004	1.846	0.034
0.18	1.849	1.872	0.012	1.796	0.029
<i>mean</i>			0.006		0.018
<i>standard deviation</i>			0.004		0.015

## Results show that simple interpolation is worse than best-fit for efficiency

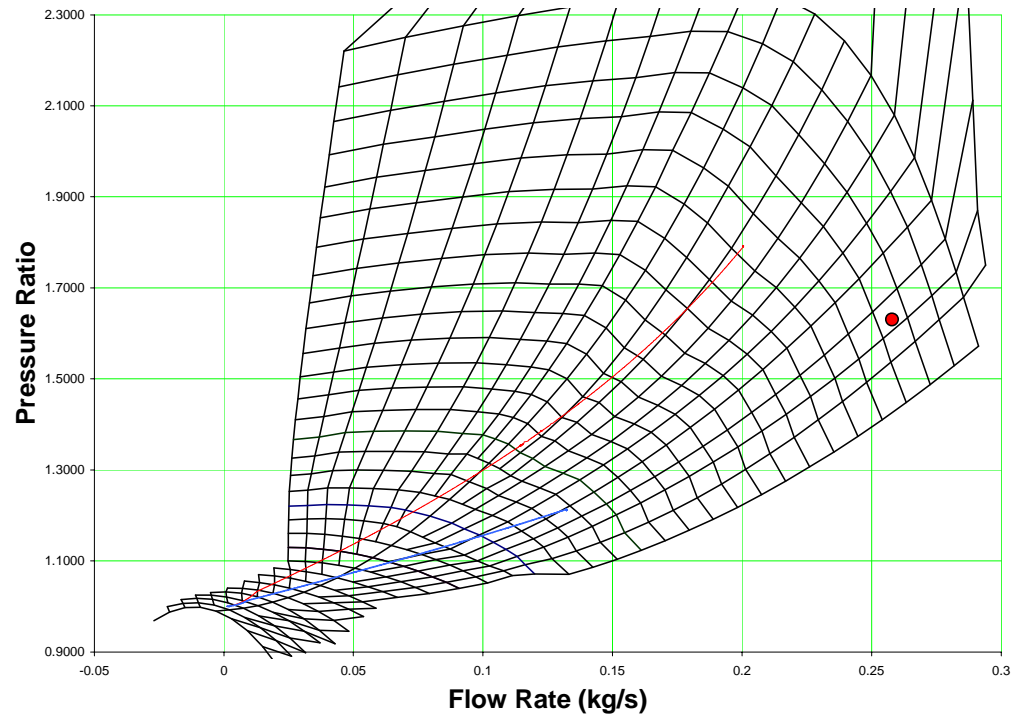
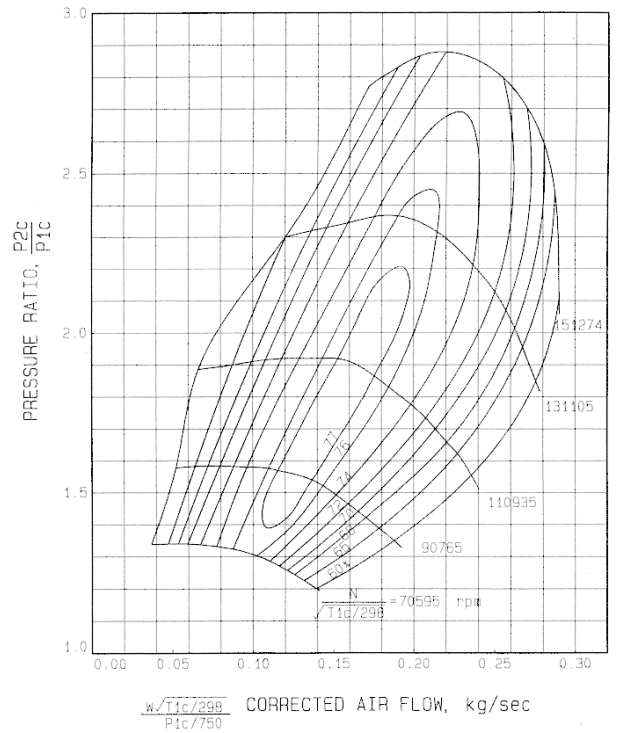
$$\text{Fractional Error} = \frac{|\text{calculated} - \text{actual}|}{\text{actual}}$$

<i>Flow Rate (kg/s)</i>	$\eta_{ad}$	<i>Interpolated</i>	<i>Fractional Error</i>	<i>Best-fit</i>	<i>Fractional Error</i>
0.12	0.744	0.711	0.044	0.719	0.034
0.14	0.736	0.727	0.012	0.739	0.004
0.16	0.77	0.719	0.066	0.746	0.031
0.18	0.765	0.693	0.094	0.737	0.037
<i>mean</i>			0.054		0.026
<i>standard deviation</i>			0.035		0.015

# Increased resolution and range of performance map based on automobile turbocharger compressor

COMPRESSOR PERFORMANCE "Small" compressor wheel  
MODEL TD05-16G 49178-05200

Test Performance Map



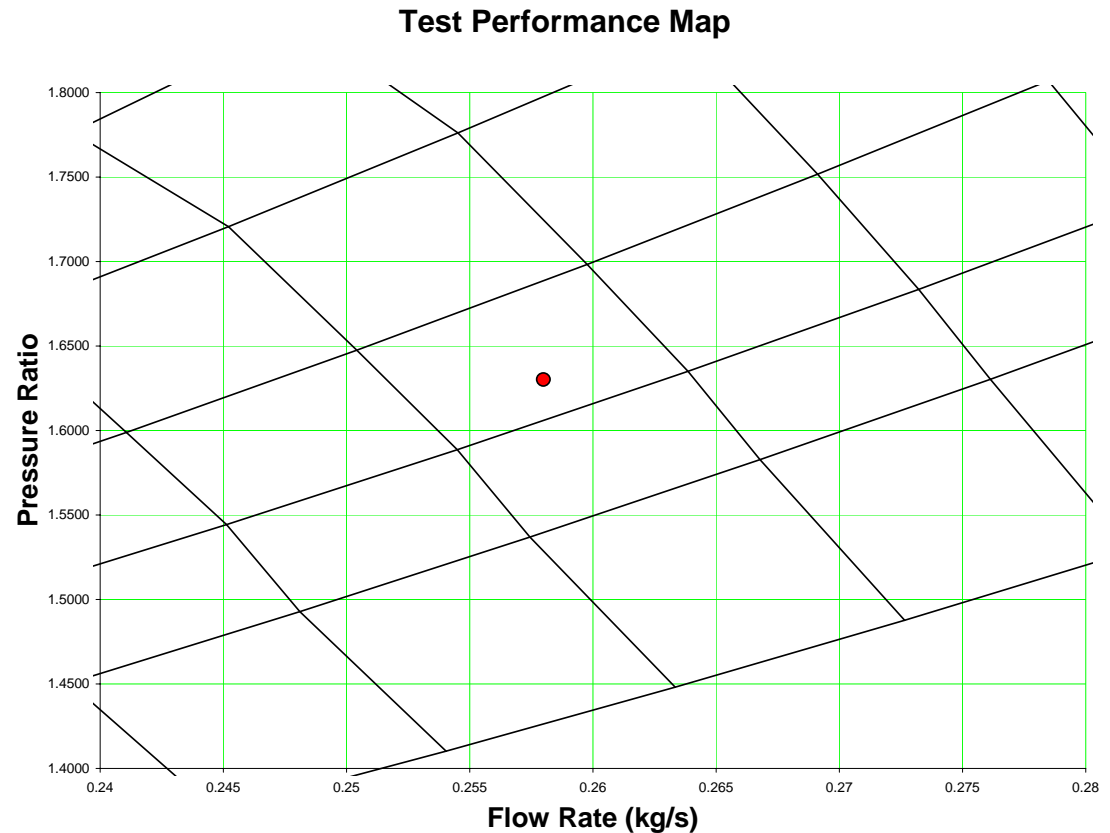
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Performance Data

With Extrapolated and Interpolated Numbers

# “Smart” interpolation based on efficiency

- Bracket with Upper and Lower speed curve
- Choose Upper and Lower Flow Values based on efficiency
  - Decide whether closer to upper or lower speed curve
    - Shift data location of further-away point
    - Done when one end-point matches
- Special cases for edges of table



## Implementation of model in the code

Total pressure

$$P_2^T = P_2 + \frac{\alpha_f \rho_f |\bar{v}_f \cdot \dot{v}_f| + \alpha_g \rho_g |\bar{v}_g \cdot \dot{v}_g|}{2} \quad (24)$$

Head

$$H = P_2^T \left( \frac{R_p - 1}{R_p} \right) \quad (25)$$

Torque

$$\tau_T = \frac{-H \cdot \dot{m}}{\omega \cdot \eta \cdot \rho} \quad (26)$$

where

$$\dot{m} = \dot{m}_J \quad \text{and} \quad \rho_{avg} = \frac{\rho_V + \rho_J}{2}$$

Dissipation

$$W_D = \tau_T \cdot \omega - H \cdot \left( \frac{\alpha_f \rho_f v_f + \alpha_f \rho_f v_f}{\alpha_f \rho_f + \alpha_f \rho_f} \right) \cdot A \quad (27)$$

# Results at Low Pressure Ratio

$$P_{in} = 90969 \text{ Pa}, T_{in} = 300 \text{ K}, R_p = 1.214, \eta = 0.647$$

	Theoretical Value	Code Value	Error
Compressor $\Delta P$ (Pa)	19452	19458	~0
Enthalpy Rise $\Delta h$ (J/kg)	17159 (isentropic) 26601 (real)	27936	5%
Power (W)	3518	3695	

## Torque (J)

Isentropic	0.3252	0.3310	1.8%
Dissipative	0.1773	0.1805	
Total	0.5025	0.5115	

Density (kg/m <sup>3</sup> )	1.1350	1.1151	1.8%
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## Power (W)

Isentropic	2267	2307	1.8%
Dissipative	1236	1258	
Total	3503	3566	
$Q_{Wall}$		3694	

# Results at High Pressure Ratio

$$P_{in} = 69371 \text{ Pa}, T_{in} = 306 \text{ K}, R_p = 1.793, \eta = 0.734$$

	Theoretical Value	Code Value	Error
Compressor $\Delta P$ (Pa)	55261	55372	0.2%
Enthalpy Rise $\Delta h$ (J/kg)	55912 (isentropic) 76333 (real)	91227	20%
Power (W)	15309	18297	

## Torque (J)

Isentropic	0.9618	1.0101	5%
Dissipative	0.3482	0.3657	
Total	1.3100	1.3758	

Density (kg/m <sup>3</sup> )	0.9919	0.9445	5%
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## Work (Watts)

Isentropic	11174	11734	5%
Dissipative	4045	4248	
Total	15218	15982	
$Q_{Wall}$		18298	



# High Pressure Ratio, No Dissipation

$$P_{in} = 69554 \text{ Pa}, T_{in} = 303.7 \text{ K}, R_p = 1.786, \eta = 1.0$$

	Theoretical Value	Code Value	Error
Enthalpy Rise $\Delta h$ (J/kg)	55876	69052	24%
Internal Energy Rise $\Delta u$ (J/kg)	49310	49450	0.3%
<i>RELAP5-3D thermal energy eqn for steady, single-phase flow</i>			
$(\dot{m}U)_j^{j+1} + \left( P \frac{\dot{m}}{\rho} \right)_j^{j+1} = \dot{Q} \Rightarrow U_2 - U_1 + P_2 \left[ \frac{1}{\rho_2} - \frac{1}{\rho_1} \right] = \frac{\dot{Q}}{\dot{m}}$			
$P_2 \left[ \frac{1}{\rho_2} - \frac{1}{\rho_1} \right]$ (J/kg)		-49443	7 J/kg
<i>Rewritten in terms of specific enthalpy, h</i>			
$h_2 - h_1 - \frac{P_1 - P_2}{\rho_1} = \frac{\dot{Q}}{\dot{m}}$			
$\left[ \frac{P_2 - P_1}{\rho_1} \right]$ (J/kg)	55876	69062	24%
$\left[ \frac{P_2 - P_1}{0.5 * (\rho_1 + \rho_2)} \right]$ (J/kg)	55876	56077	0.4%

## High Pressure Ratio, Correction to Dissipation Term

$$P_{in} = 69480 \text{ Pa}, T_{in} = 304.4 \text{ K}, R_p = 1.789, \eta = 0.733$$

	Theoretical Value	Code Value	Error
Compressor $\Delta P$ (Pa)	55078	55203	0.2%
Enthalpy Rise $\Delta h$ (J/kg)	76249	77938	2.2%

Power $[\dot{m} \cdot \Delta h]$ (W)	15229	15684	3%
$Q_{Wall}$ (W)		15682	

### Torque (J)

Isentropic	0.9578	0.9870	3%
Dissipative	0.3494	0.3601	
Total	1.3072	1.3471	

Density (kg/m <sup>3</sup> )	0.9962	0.9666	3%
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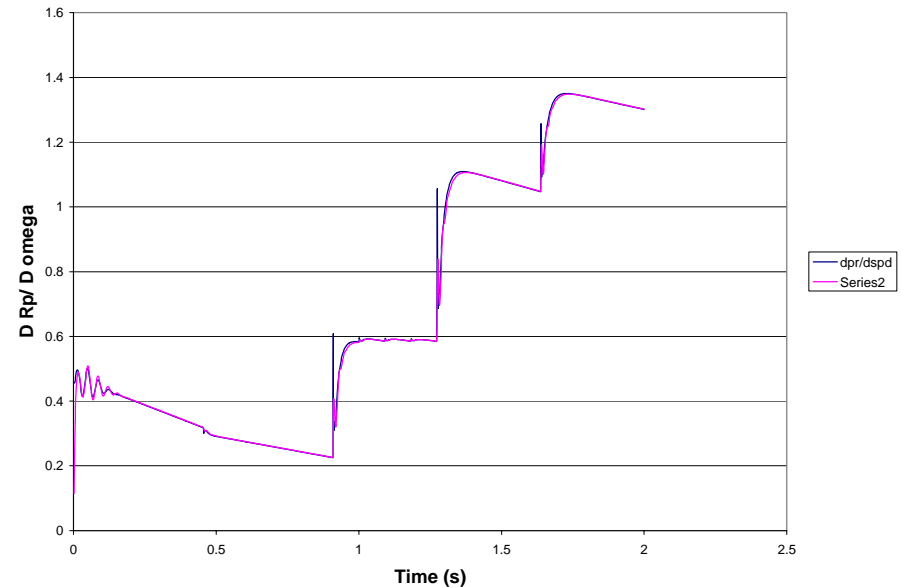
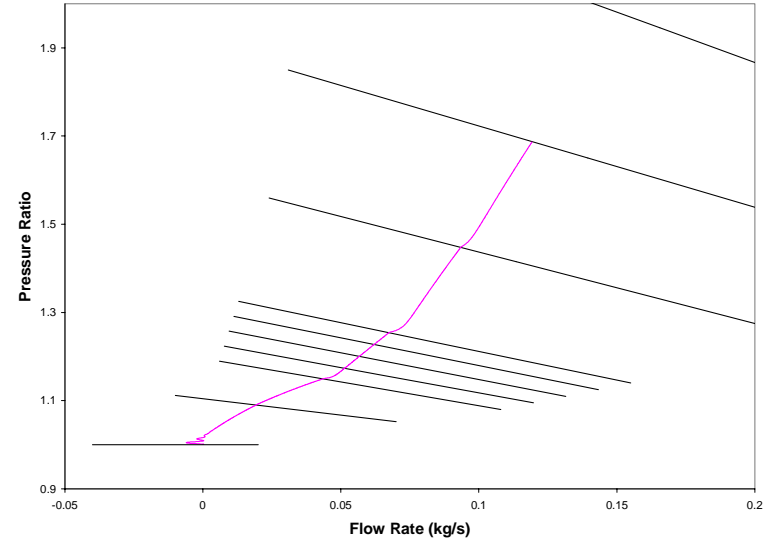
### Power (W)

Isentropic	11127	11466	3%
Dissipative	4059	4183	
Total	15186	15649	

Internal Energy Rise $\Delta u$ (J/kg)	55485	55820	0.6%
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## Linear interpolation is problematic

- Investigate instability of model when crossing node lines
- Performance map simplified to end points
- Simplified model consists of flow boundary condition and compressor with speed table.
- Ramp calculation for 2 seconds
  - Zero to 0.01 kg/s
  - Zero to Rated Speed
- Discontinuity noted in  $\Delta R_p / \Delta \omega$  at major nodes
- No discontinuity at minor nodes
- Time-averaged  $R_p$  (3 time steps) reduced severity



# ***Bicubic interpolation***

- *Method implemented but not yet functional*
  - *investigation in progress*
  - *funding suspended*
- *Requires finely identified data*
- *Feasibility questionable*
  - *may require orthogonal data*

# Remaining Items and Issues

- *Energy conservation (Resolved)*
- *Xmgr and Pygi*
- *Finalize method for data input*
  - *Performance map*
    - *Bicubic interpolator*
    - *Modified linear interpolation*
  - *Homologous*
- *Resolve performance data*
  - *Surge*
  - *Choking*
  - *Reverse flow and/or rotation*
- *Nearly-implicit*

# Conclusions

- *Compressor implementation similar to pump*
- *Performance characteristics need further analysis*
- *Additional work required to finalize model*
- *Need performance data*
- *Need transient information*